AA2016-9

AIRCRAFT ACCIDENT INVESTIGATION REPORT

CIVIL AVIATION BUREAU, MLIT J A 0 0 1 G

November 24, 2016



The objective of the investigation conducted by the Japan Transport Safety Board in accordance with the Act for Establishment of the Japan Transport Safety Board and with Annex 13 to the Convention on International Civil Aviation is to determine the causes of an accident and damage incidental to such an accident, thereby preventing future accidents and reducing damage. It is not the purpose of the investigation to apportion blame or liability.

> Kazuhiro Nakahashi Chairman, Japan Transport Safety Board

Note:

This report is a translation of the Japanese original investigation report. The text in Japanese shall prevail in the interpretation of the report.

AIRCRAFT ACCIDENT INVESTIGATION REPORT

AIRFRAME DAMAGE FROM LIGHTNING STRIKE IN FLIGHT CIVIL AVIATION BUREAU, MLIT GULFSTREAM AEROSPACE G-IV, JA001G AT AN ALTITUDE OF ABOUT 6,500 FT ABOUT 4 KM SOUTH OF NIIGATA AIRPORT

AT ABOUT 14:35 JST, MARCH 13, 2015

November 4, 2016

Adopted by the Japan Transport Safety Board

Chairman	Kazuhiro Nakahashi
Member	Toru Miyashita
Member	Toshiyuki Ishikawa
Member	Sadao Tamura
Member	Keiji Tanaka
Member	Miwa Nakanishi

1. PROCESS AND PROGRESS OF INVESTIGATION

1.1	Summary of the	On Friday, March 13, 2015, Gulfstream Aerospace G-IV,	
	Accident	registered JA001G, operated by the Civil Aviation Bureau of the	
		Ministry of Land, Infrastructure, Transport and Tourism, received a	
		lightning strike to its airframe in flight for flight inspection over the	
		vicinity of Niigata Airport, and damaged there.	
		There were five people on board the aircraft, consisting of a	
		captain and four crew members. No one was injured.	
1.2	Outline of the	On March 13, 2015, the Japan Transport Safety Board	
	Accident	designated an investigator-in-charge and two investigators on March	
	Investigation	13, 2015 to investigate this accident.	
		An accredited representative of the United States of America,	
		as the State of Design and Manufacture of the aircraft involved in	
		the accident, participated in the investigation.	
		Comments were invited from parties relevant to the cause of the	
		accident and the relevant State.	

2. FACTUAL INFORMATION

2.1	History of the	According to the statements of the captain, the first Officer and
	\mathbf{Flight}	the onboard maintenance engineer and the records of the digital flight
		data recorder, the flight history was as outlined below.
		Gulfstream Aerospace G-IV, registered JA001G, operated by the
		Civil Aviation Bureau of the Ministry of Land, Infrastructure,
		Transport and Tourism (MLIT), took off from Obihiro Airport in
		accordance with Instrument Flight Rules (IFR) at 11:16 Japan
		Standard Time (JST: UTC + 9 hrs, all times are indicated in JST on
		a 24-hour clock) on March 13, 2015, with the captain and four crew
		members on board, for the purpose of carrying out flight inspection
		affair (inspections conducted to confirm the required performance of
		aeronautical navigation facilities).
		In their pre-departure check on meteorological data, the captain
		and the first Officer were aware that the weather in Hokkaido and
		northern Japan was in a recovery trend, and that it was not in a
		situation indicating the possibility of lightning strikes or other
		significant weather on the planned flight route of the aircraft, which
		give specified would impact on the flight. In the cockpit of the aircraft,
		the captain was in charge of flying in the left seat, on the other hand
		the first officer was in charge of duties other than flying in the right
		seat.
		While the aircraft was flying south-southwest on Airway V31
		toward the Niigata VORTAC (aeronautical radio navigation aid) at an
		altitude 9,000 ft, the captain and the first officer had sighted lower
		layer of the cloud but had visually confirmed that there was no cloud
		on the flight route to the vicinity of Niigata Airport. At about 14:32, the
		aircraft commenced a descending turn from right above Niigata Airport
		in order to inspect on the next airway. At this time, the captain and the
		first officer had checked on an indication on the weather radar display
		of the aircraft showing a precipitation area, albeit not on their route,
		and were therefore focusing attention in that direction. They then
		encountered scattered thin cloud, but this was judged not to be thick
		cloud which might lead to turbulence or thunder. At about 14:35, as the
		aircraft entered the thin cloud at an altitude about 6,500 ft, a very loud
		noise was heard on the radio equipment while the captain, the first
		officer and the onboard maintenance engineer were listening to it. After
		three to four seconds, the captain and the first officer received an
		impact accompanied by a loud popping noise, together with a strong
		flash from the front of the aircraft.
		While continuing the flight, the captain and the first officer
		checked the instruments to check whether abnormality had occurred to
		the aircraft; consequently, they confirmed that discoloration was found
		on some of the cockpit instrument displays, and that the airborne DME

(Distance Measuring Equipment) display on the Captain's side had disappeared. However, there was no abnormality in the DME display on the first officer's side and there was no problem with the visibility of the discolored instrument displays; accordingly, they judged there to be no serious abnormality which would affect the flight.

The captain, the first officer and the onboard maintenance engineer discussed whether they should continue the flight inspection, and although there was thought to be no hindrance to the flight, they decided to suspend its affairs and return to the base of the aircraft at Tokyo International Airport, because they could not confirm damage to the airframe while on board. Thereafter the aircraft landed at Tokyo International Airport at 15:26. The captain and the first officer felt no abnormality when landing.

On the post-flight inspection, damage was confirmed to the left front fuselage outer skins, the front lower fuselage and the left horizontal stabilizer.



		or burnt (about 20 locations)	n fucelese sential
		damaged	r iuselage partially
		• Tail Section: Tip of static discharger on left hori	zontal stabilizar lost
		trailing edge and end of left h	orizontal stabilizer
		damaged	
		Outer skins around static discharge	er mounting surface
		of left horizontal stabilizer discolore	d and damaged
		Static disc	harger mounting surface
			Front
			Brinninge
			Jpper surface of stabilizer
		Photo I The accident aircraft Photo 2 Da	mage to left
		no	rizontai stabilizer
		Front	6.01h
		06:29	
			disappear
			25.9
		Photo 3 Location of damage to outer Photo 4 Def	ect of captain's side
		skin of left front lower fuselage dis	play(airborne DME)
2.4	Personnel	(1) Captain male, Age 42	
	Information	Airline transport pilot certificate (Airplane)	September 13, 2013
		Type rating for Gulfstream Aerospace G-IV	January 5, 2007
		Class 1 aviation medical certificate Validity:	March 11, 2016
		Specific Pilot Competence	
		Expiry of practicable period for flight	September 13, 2015
		Total flight time	4,298 hr 01 min
		Total flight time on the type of aircraft	3,082 hr 05 min
		(2) First Officer male, Age 48	
		Commercial pilot certificate (Airplane)	June 6, 1995
		Instrument flight contificate (Aimlane)	February 6, 2015
		Class 1 aviation modical cortificato Validity	$\begin{array}{c} \text{January 12, 1999} \\ \text{August 14, 2015} \end{array}$
		Snecific Pilot Competence	August 14, 2010
		Expiry of practicable period for flight	February 6 2016
		Total flight time:	5,572 hr 25 min
		Total flight time on the type of aircraft:	1,016 hr 20 min
2.5	Aircraft	(1) Aircraft type: Gulfstre	eam Aerospace G-IV
	Information	Serial number	- 1190

		Date of manufacture	July 9, 1992
		Certificate of airworthiness	No. TO-26-139
		Validity	July 21, 2015
		Category of airworthiness	Airplane Transport T
		Total flight time	9,964 hr 46 min
		(2) The aircraft was equipped with a digital fl	ight data recorder and a
		cockpit voice recorder.	
		The digital flight data recorder retained	records at the time of the
		accident; however, there were no records in	ndicating an abnormality
		related to the accident. Besides, the cocky	oit voice recorder did not
		retain any voice records during flight	since they had been
		(3) It is estimated that both the weight and th	a position of the contar of
		(3) It is estimated that both the weight and the	on the assident essured
96	Mataaralagiaal	(1) Accomputing which we approximation of (1)	igoto Airport during the
2.0	Information	(1) Aeronautical weather observations for N.	ligata Airport during the
	mormation	14:00 Wind direction: 2009 Wind velocity:	11 l . +
		14.00 Wind direction. 500 Wind velocity.	11 Kt Waathar' Light aboutors
		Cloud: amount 1/8 aumulus	aloud hass 1 000 ft
		Cloud: amount 5 / 8, cumulus	cloud base 1,000 ft
		Cloud: amount 578, cumulus	cloud base 1,500 ft
		Temperature: CC Devenint: 280	cioud base 2,500 ft
		Altimaton sotting (ONH): 20.00 in Hr	
		Attimeter setting (QINH): 29.99 Infig	07 14
		14.57 Wind direction: 500° Wind velocity.	07 KI
		Weather: Light thun denoterm	
		Cloud: amount 1/8 aumulus	aland have 1 000 ft
		Cloud: amount $f / 8$, cumulus	cloud base 1,000 It
		Cloud: amount 8/8, cumulus	cloud base 2,000 ft
		Torrest errotung: 590 Democint: 290	cloud base 2,500 It
		Altimator acting (ONU): 20 00 in Ur	
		Linktonia (linkt) shares	
		Thur denoterm is showned at 10 k	m anotherest maring to
		inunderstorm is observed at 10 k	m southeast, moving to
		(2) The weather situation around Nijesta Aim	aant waa aa fallawa
		(2) The weather situation around Nigata Air	port was as follows.
		(See the Fig. 2, Fig. 3)	on northorn Ionon was
		A winter pressure pattern centered of	of the provious day had
		begun to logger gradually, but cold air still	n of the previous day had
		The Niigate Less Material Observe	remained up in the sky.
		lightning advisorias A constant lightning advisorias	atory nau issued gale and
		See of Japon coast side and man light	aing had been detected
		besides the pessibility of lightning second	ing nau been detected,
		was reported in the terminal accordance for	ng near Nigata Airport
		was reported in the terminal aerodrome for	ecast (1111).



*1 The thunder nowcast is one of the Japan Meteorological Agency forecast system, and it analyzes the violence of the thunder and a possibility of the thunder by 1 km grid unit, predicts it until the one hour later (ten minutes – 60 minutes destination) renews every ten minutes and offers it.

3. ANALYSIS

3.1	Involvement of	Yes
	Weather	
3.2	Involvement of	None
	Pilots	
3.3	Involvement of	None
	Aircraft	
3.4	Analysis of	(1) Time and location of lightning strike
	Findings	It is highly probable that the aircraft was struck by lightning
		at an altitude of about 6,500 ft about 4 km south of Niigata Airport
		at about 14:35 while commencing a descending turn after passing
		over the airport.
		(2) Involvement of weather
		It is highly probable that electrically charged cumulus clouds
		were distributed over a wide area at low altitude along the flight
		route of the aircraft, since observed information shows that the
		weather situation near the airport on the day of the accident was
		a winter pressure pattern with high pressure in the west and low
		in the east; moreover, because cold air still remained up in the sky
		around the time when strong winter pressure pattern of the
		previous day had begun to loosen gradually.
		It is probable that the lightning strike that damaged the aircraft
		had the properties of wintry lightning caused by cumulus clouds
		distributed over a wide area at low altitude, which frequently occur
		on the Sea of Japan side in winter, unlike localized cumulonimbus
		clouds that frequently occur in the summer.
		It is probable that, although the captain and the first officer had
		entered into scattered thin cloud while commencing a descending
		turn and were aware that a precipitation area was indicated on the
		airborne weather radar display, the aircraft came closer to the
		electrically charged cumulus clouds and encountered lightning
		strike in a situation in which it was difficult to predict and avoid
		lightning strike in the scattered thin cloud. In addition, just before
		an impact due to the lightning strike, the captain, the first officer
		and the onboard maintenance engineer heard noise with the radio
		equipment they were listening to, but it is probable that this was
		due to the effect of the lightning.
		(3) Cause of damage
		The damage to the outer skin of left front lower fuselage and
		the end of left horizontal stabilizer, the burn out of the rivets and
		the loss of the static discharger are commonly found in aircraft
		having struck by lightning. Therefore, it is probable that when the
		lightning strike, a lightning discharge path was formed from the
		left front of the fuselage to the left horizontal stabilizer and the
		static discharger. In addition, it is probable that the malfunction of

the cockpit instrument display was affected by the fact that
locations impacted by the lightning strike were close to the cockpit.

4. PROBABLE CAUSES

It is probable that this accident occurred because the aircraft sustained damage to the outer skins of its left front lower fuselage as a result of being struck by lightning to the airframe in flight.

It is probable that the lightning strike was occurred because the aircraft came close to the electrically charged cumulus clouds in the scattered thin cloud where it was difficult to predict lightning.